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SEQ	Task Card / EO	Task Card / EO Description	Certify	NOTE
1	08-00-00-M01	<p>AIRCRAFT WEIGHING Carry out aircraft weighing.</p> <p>NOTE: Total weight changes more than +/- 190 kg or change of the center of gravity higher than 19 mm, as a result of modifications, require a Weight and Center of Gravity Change Sheet to be issued and aircraft weighing to be performed.</p> <p>NOTE: Interval is 1461 dy.</p>		
2	20-Z140-010	<p>EWIS COMPONENTS OF GENERATOR POWER HARNESS General Visual Inspection of EWIS components of Generator Power Harness at Center E-Bay for signs of damage, adequate installation, chafing and general condition. EWIS components are located at center fuselage II center E-BAY area and AFT pressure bulkhead AFT face harness area - INTERNAL.</p> <p>INTERVAL: 24000 FH NOTE: EWIS.</p>		
3	20-Z220-003	<p>EWIS COMPONENTS OF COCKPIT CONSOLE, PEDALS AND SEATS HARNESS General Visual Inspection of EWIS components of Cockpit Console, Pedals and Seats Harness inside of Cockpit for signs of damage, adequate installation, chafing and general condition.</p> <p>INTERVAL: 24000 FH NOTE: EWIS.</p>		
4	20-Z220-010	<p>COCKPIT LH AND RH CONSOLE Cleaning of Cockpit LH and RH console area.</p> <p>INTERVAL: 24000 FH NOTE: EWIS.</p>		
5	20-Z220-011	<p>EWIS COMPONENTS OF COCKPIT LH AND RH CONSOLE HARNESS General Visual Inspection of EWIS components of LH and RH CB panels harness inside of Cockpit for signs of damage, adequate installation, chafing and general condition.</p> <p>INTERVAL: 24000 FH NOTE: EWIS.</p>		



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6	20-Z220-014	FORWARD FUSELAGE SIDE TOP AREAS, INCLUDING AREA COVERED BY GALLEY, TOILET AND WARDROBE - INTERNAL Cleaning of Fwd Fuselage Side Areas Near Door (from Fr12/13 to Fr 20/23) at area covered by Toilet , Wardrobe and Galley. INTERVAL: 24000 FH NOTE: EWIS.		
7	20-Z230-001	RST - CENTER FUSELAGE I SIDE AREAS, BEHIND DADO PANELS - INTERNAL Cleaning of Center Fuselage I Side Areas behind Dado Panels, from Forward Fuselage and Center Fuselage I Junction to Center Fuselage I and II Junction - Internal side of Fuselage. INTERVAL: 12000 FH NOTE: EWIS.		
8	20-Z240-001	RST - CENTER FUSELAGE II SIDE AREAS, BEHIND DADO PANELS - INTERNAL Cleaning of Center Fuselage II Side Areas behind Dado Panels, from Center Fuselage I and II Junction to Center Fuselage II and III Junction - Internal Side of Fuselage. INTERVAL: 12000 FH NOTE: EWIS.		
9	20-Z250-001	RST - CENTER FUSELAGE III SIDE AREAS, BEHIND DADO PANELS - INTERNAL Cleaning of Center Fuselage III Side Areas behind Dado Panels, from Center Fuselage II and III Junction to Region Ahead Galley / Toilet - Internal Side of Fuselage. INTERVAL: 12000 FH NOTE: EWIS.		
10	20-Z250-004	EWIS COMPONENTS OF EPD HARNESS AT AFT GALLEY AND TOILET AREA General Visual Inspection of EWIS components of Electrical Power Distribution Harness at aft galley and toilet area for signs of damage, adequate installation, chafing and general condition. INTERVAL: 24000 FH NOTE: EWIS.		
11	20-Z500-002	EWIS COMPONENTS OF WING TRAILING EDGE HARNESS General Visual Inspection of EWIS components of Wing Trailing Edge Harness for signs of damage, adequate installation, chafing and general condition. INTERVAL: 24000 FH NOTE: EWIS.		



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12	20-Z500-006	EWIS COMPONENTS OF WING MAIN BOX AREA General Visual Inspection of EWIS components of Wing Main Box Area from Rib 23 to Wing Tip Rib (Dry Area) Harness for signs of damage, adequate installation, chafing and general condition. INTERVAL: 24000 FH NOTE: EWIS.		
13	21-25-03-001C	OPC - EMERGENCY RAM AIR CHECK VALVE Operational Check of Emergency Ram Air Check Valve. NOTE: Interval is 24000 fh. NOTE: CMR two star.		
14	21-32-00-001C	FNC - CABIN PRESSURE RELIEF COMPONENTS Functional Check of positive pressure relief setting of Positive Pressure Relief Valve. NOTE: Interval is 24000 fh. NOTE: CMR two star.		
15	21-51-06-001	CONDENSER REHEATER Cleaning of Condenser Reheater core.		
16	25-01-250-H001	AFT ELECTRONICS COMPARTMENT EMI SHIELD AND BONDING JUMPERS Aft-Electronic-Compartment EMI-Shield and Bonding Jumpers - General Visual Inspection		
17	25-11-01-002	PILOT/CO-PILOT SEATS Detailed Inspection of Pilot/Co-Pilot Seats looking for possible degradation, cracks, damage, scratches and wear. NOTE: Interval changed from 6000 FH or 5000 FC Whichever occurs first, into 13750 fh, CMP Rev.11.		
18	27-12-01-005	AILERON PCU EXTERNAL LINKAGES Operational Check of Aileron PCUs External Linkages.		
19	27-41-01-005	HORIZONTAL STABILIZER ACTUATOR NO-BACK MECHANISM Detailed Inspection of Horizontal Stabilizer Actuator No-Back Mechanism for broken or damaged components. NOTE: Interval changed from 9000 fh into 15000 fh, CMP Rev. 11.		
20	27-51-01-003C	FNC - INBOARD FLAP ACTUATOR Functional Check of integrity of Inboard Flap Actuator. NOTE: Interval is 25000 fh. NOTE: CMR one star.		



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21	27-51-01-004C	FLAP ACTUATOR TORQUE LIMITER Flap Actuator Torque Limiter - Operational Test		
22	28-11-07-001	FNC - FUEL TANK GROUNDING POINTS Functional Check of Fuel Tanks Grounding Points. Resistance check. NOTE: Interval is 10000 fh.		
23	29-12-00-005C	FNC - NO.3 HYDRAULIC SYSTEM INTERNAL LEAKAGE Functional Check of No.3 Hydraulic System Internal Leakage. NOTE: Interval is 13000 fh. NOTE: CMR one star.		
24	29-33-02-003C	FNC - HYDRAULIC SYSTEM NO.3 TEMPERATURE SWITCH Functional Check of No.3 Hydraulic System Temperature Switch. To be complied with replacement serviceable part. NOTE: Interval is 12600 fh. NOTE: CMR one star.		
25	32-21-007-0002	NOSE LANDING GEAR WHEEL AXLE - INTERNAL Detailed Inspection of NLG Wheel Axle, LH/RH - Internal Side of Axle. NOTE: Wheels removal is not required. NOTE: Interval is 2191		
26	36-11-06-001	TORQUE MOTOR CONTROLLER FILTER Cleaning of Torque Motor Controller Wire Mesh Filter by ultrasonic method.		
27	38-41-00-001	AIR MANIFOLD AIR FILTER Discard of Air Manifold Air Filter. NOTE: Interval is 6000 fh.		
28	52-30-003-1084	FORWARD CARGO DOOR PIANO HINGE - EXTERNAL Detailed Inspection of Fwd Cargo Door Piano Hinge area at Cargo Door and Fuselage - External Side of Door. NOTE: TRSHOLD=72 MO; INTERVAL 34 MO. NOTE: Interval changed from From: "T: 72 MO I: 34 MO" To: "T: 10000 I: 10000 FC", CMP Rev.11		
29	52-30-003-1085	AFT CARGO DOOR PIANO HINGE - EXTERNAL Detailed Inspection of Aft Cargo Door Piano Hinge area at Cargo Door and Fuselage - External Side of Door. NOTE: TRESHOLD=72 MO; INTERVAL=34 MO NOTE: Interval changed from From: "T: 72 MO I: 34 MO" To: "T: 10000 I: 10000 FC", CMP Rev.11		



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30	52-43-002-1180	FORWARD AVIONICS COMPARTMENT ACCESS HATCH STOPS AT HATCH AND FUSELAGE - INTERNAL Detailed Inspection of Fwd Avionics Compartment Access Hatch Stops at Hatch and Fuselage - Internal Side of Fuselage and at Door Surrounding. NOTE: TRESHOLD=18000 FC; INTERVAL 18000 FC.		
31	53-01-240-H001	MID AVIONICS COMPARTMENT EMI SHIELD General Visual Inspection of EMI Shielding and attachment points located at Cabin Floor Panels.		
32	53-10-002-0921	FWD AVIONICS COMPARTMENT ACCESS HATCH CUTOUT STRUCTURE - EXTERNAL General Visual Inspection of Fwd Avionics Compartment Access Hatch Cutout Structure, from FR 4 to FR 6 and from STGR 25 LH to STGR 25 RH - External Side of Fuselage.		
33	53-10-008-0002	FWD PASSENGER DOOR CUTOUT STRUCTURE - INTERNAL Detailed Inspection of Fwd Passenger Door Cutout Structure (FR 16 and FR 19) around of Door Stop Attachments - Internal Side of Fuselage. I: 14545 FC		
34	53-10-009-0002	FWD SERVICE DOOR CUTOUT STRUCTURE - INTERNAL Detailed Inspection of Fwd Service Door Cutout Structure (FR 14 and FR 17) around of Door Stop Attachments - Internal Side of Fuselage I: 14545 FC		
35	53-10-012-0936	RAT COMPARTMENT CUTOUT STRUCTURE - EXTERNAL General Visual Inspection of Ram Air Turbine (RAT) Compartment Cutout Structure - External Side of Fuselage.		
36	53-10-014-0002	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Detailed Inspection of Upper Skin Panel around of Central Post and Upper Beam Attachment Holes - External Side of Fuselage. T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first		
37	53-10-014-0003	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Detailed Inspection of Upper Skin Panel around of Central Post and Lower Beam Attachment Holes - External Side of Fuselage. NOTE: Applicable for PRE-MOD SB 190-53-0041: T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first.		



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38	53-10-014-0004	<p>FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Special Detailed Inspection (Eddy Current) of Upper Skin Panel around of Central Post and Lower Beam Attachment Holes to verify the Doubler located under the Skin Panel - External Side of Fuselage.</p> <p>T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first</p> <p>NOTE: NDT inspection was performed by _____.</p> <p>Result of the NDT inspection: _____</p> <p>NDT inspector sign _____</p>		
39	53-10-016-0946	<p>FORWARD FUSELAGE LOWER SKIN PANEL - EXTERNAL General Visual Inspection of Lower Skin Panel, From Str 17 LH to Str 17 RHand From FR 1 to FR 20/23 - External Side of Fuselage.</p> <p>T: 9000 FC I: 9000 FC</p>		
40	53-10-021-0001	<p>NOSE LANDING GEAR WHEELWELL METALLIC STRUCTURE - EXTERNAL Nose Landing Gear Wheelwell Metallic Structure - External</p> <p>NOTE1: CIC reapplication is required. NOTE2: TRESHOLD=72 MO; INTERVAL=44 MO.</p>		
41	53-10-024-0931	<p>NOSE LANDING GEAR SHOCK STRUT AND UPPER DRAG BRACE HOUSING - EXTERNAL Detailed Inspection of NLG Shock Strut and Upper Drag Brace Housing - External Side of Structure.</p> <p>NOTE: Interval and Threshold changed from 6000 fc into 12000 fc, CMP Rev.11</p>		
42	53-10-025-0929	<p>NOSE LANDING GEAR WHEEL WELL CUTOUT STRUCTURE - EXTERNAL General Visual Inspection of Nose Landing Gear Wheel Well Cutout Structure - External Side of Fuselage.</p> <p>I: 10000 FC</p>		
43	53-21-003-1052	<p>CENTER FUSELAGE I LOWER SKIN PANEL - EXTERNAL General Visual Inspection of Lower Skin Panel, From Str 17R to Str 17L and From FR 20 to FR 34/35 - External Side of Fuselage.</p> <p>T: 9000 FC I: 9000 FC</p>		
44	53-22-003-1164	<p>CENTER FUSELAGE II LOWER SKIN PANEL - EXTERNAL General Visual Inspection of Lower Skin Panels, from Str 17 RH to Str 17 LH and from Fr 34 to Fr 73/74 - External Side of Fuselage.</p> <p>TRESHOLD: 9000 FC INTERVAL: 9000 FC</p>		



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45	53-22-010-1182	CENTER FUSELAGE II LONGITUDINAL SPLICES - EXTERNAL General Visual Inspection of Longitudinal Splices, Joining the Upper and Side Skin Panels - External Side of Fuselage. T: 10000 FC I: 10000 FC.		
46	53-22-011-0969	CENTER FUSELAGE II TO WING STUB SPLICES - INTERNAL Detailed Inspection of Center Fuselage II to Wing Stub Splices, from FR 45 to FR 56 below STGR 17 LH/RH - Area covered by Wing to Fuselage Fairing. NOTE: Interval is 3652 Days.		
47	53-22-011-0970	CENTER FUSELAGE II TO WING STUB SPLICES - INTERNAL Detailed Inspection of Center Fuselage II to Wing Stub Splices, from FR 45 to FR 56, below STGR 17 LH/RH - Internal Side of Fuselage. NOTE: Interval is 3652 Days.		
48	53-22-015-1169	CENTER FUSELAGE II CIRCUMFERENTIAL SKIN SPLICES AND STRINGER SPLICES - EXTERNAL General Visual Inspection of Circumferential Skin Splices and Stringer Splices, at Fr 42/43 and at Fr 58/60, Upper Portion (from Str 16 LH to Str 16 RH) - External Side of Fuselage. T: 10000 FC I: 10000 FC		
49	53-23-003-1067	CENTER FUSELAGE III LOWER SKIN PANEL - EXTERNAL General Visual Inspection of Lower Skin Panel from Str 17 LH to Str 17 RH and from FR 73 to FR 90/91 - External Side of Fuselage. T: 9000 FC I: 9000 FC		
50	53-30-015-1130	REAR FUSELAGE MACHINED FRAME 100 - INTERNAL General Visual Inspection of Machined Frame 100 - Internal Side of Fuselage. NOTE: THRESHOLD=18000 FC; INTERVAL=18000 FC		
51	55-40-005-1146	DET - RUDDER FRONT SPAR BALANCE MASS FITTINGS - INTERNAL Detailed Inspection of Rudder Front Spar Balance Mass Fittings - Internal Side of Trailing Edge. NOTE: Rudder has to be removed.. INTERVAL = 2922 Days		
52	56-12-00-001	COCKPIT DIRECT-VISION WINDOW COMPONENTS Detailed Inspection of Cockpit Direct-Vision Window Actuation and Locking Mechanism for integrity, surface corrosion, cracks, or missing items.		



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53	56-12-00-002	COCKPIT DIRECT-VISION WINDOW COMPONENTS Lubrication of Cockpit Direct-Vision Window Actuation and Locking Mechanism.		
54	57-30-001-1274	WINGLET-TO-WING SPAR ATTACHMENTS - INTERNAL General Visual Inspection of Winglet-to-Wing Spar Attachments, LH/RH - Internal Side of Winglet. NOTE: THRESHOLD = 9333 FC , INTERVAL = 9333 FC.		
55	57-50-003-1284	DET - WING SPAR 3 - EXTERNAL Detailed Inspection of Wing Spar 3 Aft Face, including Web and Caps, from Rib 1 to Rib 10, LH/RH - External Side of Wing Fixed Trailing Edge. NOTE: Flaps have to be fully extended.		
56	57-50-005-0003	WING FIXED TRAILING EDGE RIB 9A - INTERNAL Detailed Inspection of Wing Fixed Trailing Edge Rib 9A Web, LH/RH - Inside Wing Trailing Edge Box. INTERVAL: - 10000 fc and 13300 fh, applicable for PRE-MOD SB 57-0028; or - 14228 fc and 18923 fh for E195 POST-MOD SB190-57-0028; - T: 21826 fc and 29026 fh; I: 15433 fc and 20525 fh for E190 POST-MOD SB190-57-0028. NOTE: Aircraft is PRE-MOD SB 57-0028		
57	57-50-007-0008	SDI - WING FIXED TRAILING EDGE RIB 05 DOUBLER - EXTERNAL Special Detailed Inspection (Eddy-Current) of Lateral Side of Ti doubler between Rib 05 Lower Flange and Wing Lower Skin Panel, LH/RH - External Side at MLG Compartment. NOTE: Task applicable to POST-MOD SB 190-57-0031 TRESHOLD: 20321 FC OR 27027 FH INTERVAL: 7241FC OR 9631 FH		
58	57-50-024-0001	INBOARD FLAP TRACK 1 SUPPORT - INTERNAL Detailed Inspection of Flap Track Support 1, LH/RH - Internal Side of Flap Root Fairing.		
59	EO-AOA-25-25	ARM-DISARM DOOR ASSEMBLY-HINGE, AFT PASSENGER DOOR - REPLACEMENT Perform replacement of hinge on Arm-Disarm door assembly - AFT Passenger door with new reinforced hinge iaw AMM 25-23-22-000/400-801-A. NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book.		



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60	EO-AOA-27-13	<p>FLIGHT CONTROLS - DISCONNECT HANDLE - PROTECTOR PLATE INSTALLATION</p> <p>Perform modification of Aileron/Elevator Disconnect Handle iaw SB 190-27-0040 3. Accomplishment Instructions - latest revision.</p> <p>NOTE for Planning: After accomplishment of this EO, make a record in a/c log book, modification log book, SB status list and inform Embraer through to Reliability Report.</p>		
61	EO-AOA-32-40	<p>REPLACEMENT OF THE SUPPORT FLANGE OF THE SWIVEL JOINTS</p> <p>1. Prepare the aircraft iaw AMM 32-31-05-000-801-A.</p> <p>2. a) LH MLG: - Disconnect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-03A item 320) iaw AMM 20-10-03-000/400-801-A. - Replace BONDING SUPPORT PN: 190-70161-001 with PN: 190-70197-001 (AIPC 32-11-05-03A item 380A) iaw AMM 20-10-01-910-801-A. - Connect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-03A item 320A) iaw AMM 20-10-03-000/400-801-A and SNL 190-32-0037.</p> <p>2. b) RH MLG: - Disconnect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-05A item 320) iaw AMM 20-10-03-000/400-801-A. - Replace BONDING SUPPORT PN: 190-70161-001 with PN: 190-70197-001 (AIPC 32-11-05-05A item 380A) iaw AMM 20-10-01-910-801-A. - Connect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-05A item 320A) iaw AMM 20-10-03-000/400-801-A and SNL 190-32-0037.</p> <p>3. Close the job iaw AMM 32-31-05-400-801-A.</p> <p>Note: Lifting the aircraft on jacks is required. Bleeding of the hydraulic system No. 2 is required. An operational test of the landing gear is required.</p>		
62	EO-AOA-32-47	<p>LANDING GEAR - NLG DOWNLOCK SPRING SUPPORT INSTALLATION - MODIFICATION</p> <p>Perform installation of NLG Downlock spring support iaw SB 190-32-0078 3. Accomplishment Instructions.</p> <p>NOTE for CAMO: After accomplishment of this EO make a record in : - A/C log book - Mod. log book - SB digital status list - Inform Embraer</p>		



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63	EO-AOA-38-03	<p>WASTE LINE - NEW SUPPORT INSTALLATION - MODIFICATION. Perform installation of new support on waste line iaw SB 190-38-0021, 3. Accomplishment Instructions - latest revision.</p> <p>NOTE for Planning: The due date is flexible and may be prolonged if needed.</p> <p>NOTE for Engineering: After accomplishment of EO, make a record in: a/c log book. mod. log book, SB status list and inform Embraer through Reliability report.</p>		
64	EO-AOA-53-29	<p>EXTERNAL EDDY CURRENT INSPECTION OF PERMANENT REPAIR ON CENTER FSLG III SKIN Perform an external Eddy Current Inspection of permanent repair on center fuselage III skin between FR 81 - FR 82 and between Stgr 13L - 16L in accordance with NDT PART1 - 51-00-00-916-803.</p>		
65	EO-AOA-53-30	<p>EXTERNAL EDDY CURRENT INSPECTION OF PERMANENT REPAIR ON THE CENTER FUSELAGE III SKIN BETWEEN FR 89-90 AND STR 10L-12L IAW ETD 2014-190-011938 AND ADAL ITEM 70. Perform external Eddy Curent Inspection of permanent repair on the center fuselage III skin,between FR 89-90 and STR 10L-12L in accordance with SRM II 53-23-01 RE64P 5A Table 202 and NDT PART1.</p>		
66	EO-AOA-53-49	<p>FUSELAGE - REINFORCEMENT OF FRAMES OF THE FWD FUSELAGE CLOSE TO NLG BAY-MODIFICATION. Perform modification of Reinforcement of frames of the FWD fuselage close to NLG bay iaw SB 190-53-0026-3. Accomplishment Instructions.</p> <p>NOTE: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After performed of this EO, make a record in Modification log book, SB status list.</p>		
67	EO-AOA-53-50	<p>FUSELAGE-REWORK TO FWD FUSELAGE AT FRAME 8 - MODIFICATION. Perform modification of FWD fuselage by reinforcing at frame 8 iaw SB 190-53-0031-3. Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of modification, make a record in MODification log book and SB status list.</p>		



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68	EO-AOA-53-51	<p>FUSELAGE - REINFORCEMENT OF FWD FUSELAGE FRAME 11-MODIFICATION Perform modification of FWD Fuselage by reinforcement of Frame 11 iaw SB 190-53-0033-3. Accomplishemnt Instructions.</p> <p>NOTE: To perform this modification need to be open RAT Compartment door.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
69	EO-AOA-53-53	<p>FORWARD FUSELAGE - REPLACEMENT OF FASTENER IN FRAME 7 - MODIFICATION. Perform modification of fastener in Frame 7 iaw SB 190-53-0043 R02-3. Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
70	EO-AOA-53-54	<p>FORWARD FUSELAGE - REINFORCEMENT OF NLG BULKHEAD WEB AND FRAME 8 - MODIFICATION. Perform modification of reinforcement of NLG Bulkhead web and frame 8 iaw SB 190-53-0049 R02-3. Accomplishment Instructions, SB 190-53-0049 IN1 and SB 190-53-0049 IN2.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
71	EO-AOA-53-55	<p>FORWARD FUSELAGE - LATERAL POST- T SHAPE REPLACEMENT- MODIFICATION. Perform modification of reinforcement of T shape lateral post iaw SB 190-53-0050 R01-3. Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
72	EO-AOA-53-56	<p>FUSELAGE - STRUCTURAL REINFORCEMENT OF LOWER BEAM IN THE COCKPIT WINDSHIELD STRUCTURE- MODIFICATION. Perform modification of Structural Reinforcement of Lower Beam in the Cockpit Windshield Structure iaw SB 190-53-0053-3. Accomplishment Instructions and SB 190-53-0053 IN1.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		



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73	EO-AOA-53-57	<p>ANCHOR NUTS AND FASTENTENERS OF BOTH WINDSHIELD STRUCTURE-GENERAL VISUAL INSPECTION.</p> <p>1. Perform GVI of both WIndshield Anchor nuts and attaching fasteners.</p> <p>2. If found any discrepancy make record in Work Card and perform rectifying action.</p> <p>NOTE: Sheet metal specialist will be needed ifthere is rectifying actions.</p> <p>NOTE: For required inspection windshields must be removed.</p>		
74	EO-AOA-53-58	<p>FORWARD FUSELAGE - REINFORCEMENT OF DIRECT VISION WINDOWS- MODIFICATION.</p> <p>Perform modification of reinforcement of T shape lateral post iaw SB 190-53-0052-3. Accomplishment Instructions, SB 190-53-0052 IN 1 and SB 190-53-0052 IN 2.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p> <p>NOTE for CAMO: Recommended to be performed before TCN: 20000 FC.</p>		
75	EO-AOA-53-60	<p>FUSELAGE-WING-TO-FUSELAGE FAIRING AFT PANEL-FLAP MECHANISM ACCESS DOOR - ELECTRICAL BONDING IMPROVEMENT - MODIFICATION.</p> <p>Perform modification of Wing-to-Fuselage fairing AFT panel-Flap mechanism access door iaw SB 190-53-0091 3. Accomplishment Instructions, SB 190-53-0091 IN1, SB 190-53-0091 IN2 and SB 190-53-0091 IN3.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB Status list.</p>		
76	EO-AOA-57-14	<p>WINGS - STRUCTURAL REINFORCEMENT OF THE FIXED TRAILING EDGE RIB 9A-MODIFICATION.</p> <p>Perform modification of Structural Reinforcement of the Fixed Trailing Edge Rib 9A iaw SB 190-57-0028-3. Accomplishment Instructions PART I nad PART II and SB 190-57-0028 IN 3.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		



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77	EO-AOA-57-16	<p>WINGS - STRUCTURAL REINFORCEMENT OF WING STUB SIDE FITTING - MODIFICATION. Perform modification of reinforcement of wing stub side fitting iaw SB 190-57-0046. 3.Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p> <p>NOTE for CAMO: Recommended to be performed before TCN: 20000 FC.</p>		
78	EO-AOA-57-17	<p>WINGS - REINFORCEMENT OF FIXED TRAILING EDGE RIB 5 AND TRUNNION - MODIFICATION. Perform modification of reinforcement of fixed trailing edge rib 5 and trunnion iaw SB 190-57-0051 . 3.Accomplishment Instructions PART I and PART II.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p> <p>NOTE for CAMO: Recommended to be performed before TCN: 20000 FC.</p>		
79	Z220-002	<p>GVI - COCKPIT LH AND RH CONSOLE, COCKPIT BULKHEAD, AND COCKPIT DOOR - INTERNAL General Visual Inspection of Cockpit LH/RH Console Area, Cockpit Bulkhead and Cockpit Door - Internal Side of Cockpit. NOTE: EWIS.</p> <p>INTERVAL: 24000 FH</p>		
80	Z220-004	<p>FORWARD FUSELAGE FRONT AREA (INSIDE COCKPIT), FORWARD PARTITION AND FLOOR AREA - INTERNAL General Visual Inspection of Cockpit Fwd Partition and Floor Areas - Internal Side of Cockpit.</p>		
81	Z230-001	<p>GVI - CENTER FUSELAGE I SIDE AND TOP AREAS - INTERNAL General Visual Inspection of Center Fuselage I Side and Top Areas, from Forward Fuselage and Center Fuselage I Junction to Center Fuselage I and II Junction, including Floor Structure - Internal side of Fuselage.</p> <p>NOTE: EWIS. INTERVAL: 24000 FH</p>		
82	Z240-001	<p>GVI - CENTER FUSELAGE II, SIDE AND TOP AREAS - INTERNAL General Visual Inspection of Center Fuselage II Side and Top Areas, from Center Fuselage I and II Junction to Center Fuselage II and III Junction - Internal Side of Fuselage. NOTE: EWIS. INTERVAL: 24000 FH</p>		



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83	Z240-003	GVI - CENTER FUSELAGE II, REGION BETWEEN WING STUB AND FLOOR PANELS- INTERNAL General Visual Inspection of Center Fuselage II, region between Wing Stub and Floor Panels, including Floor Structure - Internal Side of Fuselage NOTE: EWIS. INTERVAL: 24000 FH		
84	Z250-001	GVI - CENTER- FUSELAGE III SIDE AND TOP AREAS, EXCEPT REGION FROM GALLEY/TOILET TO REAR PRESSURE BULKHEAD - INTERNAL General Visual Inspection of Center Fuselage III Side and Top Areas, from Center Fuselage II and III Junction to Region Ahead of Galley / Toilet - Internal Side of Fuselage. NOTE: EWIS. Galley and Toilet removal is not required.		
85	Z300-002	EXTERNAL POWER AND HYDRAULIC SYS 3 GROUND CONNECTION, JACKING POINT - INTERNAL General Visual Inspection of External Power, No. 3 Hydraulic System Ground Connection and Jacking Point - Internal Side of Compartments. INTERVAL: 24000 FH		
86	Z419-001	PYLON AND WING LOWER SKIN PANEL COVERED BY PYLON FAIRING - INTERNAL General Visual Inspection of LH/RH Pylon and Pylon Fairings, Including the Area of Wing Lower Skin Covered by Pylon Fairings - Internal Side of Pylon and Fairings. NOTE: EWIS. INTERVAL: 12000 FH		