



WORKPACK ACCOUNTABILITY FORM

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SEQ	Task Card / EO	Task Card / EO Description	Certify	NOTE
1	52-30-001-0445	<p>FORWARD CARGO DOOR STRUCTURE, HANDLE BOX AND VENT VALVE - INTERNAL General Visual Inspection of Fwd Cargo Door Structure, including Handle Box and Vent Valve - Internal Side of Door.</p> <p>NOTE: TRESHOLD 72 MO; INTERVAL72 MO = (2191 dy)</p>		
2	52-30-001-0446	<p>AFT CARGO DOOR STRUCTURE, HANDLE BOX AND VENT VALVE - INTERNAL General Visual Inspection of Aft Cargo Door Structure, including Handle Box and Vent Valve - Internal Side of Door.</p> <p>NOTE: TRESHOLD=72 MO; INTERVAL=72 MO = (2191 dy)</p>		
3	52-30-005-1089	<p>FORWARD CARGO DOOR LATCH SHAFTS AND LATCH SHAFT FITTINGS - INTERNAL Detailed Inspection of Fwd Cargo Door Latch Shafts and Latch Shaft Fittings - Internal Side of Door.</p> <p>NOTE: Interval changed from 58 MO into 72 MO (=2191 dy), CMP Rev.11.</p>		
4	52-30-005-1090	<p>AFT CARGO DOOR LATCH SHAFTS AND LATCH SHAFT FITTINGS - INTERNAL Detailed Inspection of Aft Cargo Door Latch Shafts and Latch Shaft Fittings - Internal Side of Door.</p> <p>NOTE: Interval changed from 58 MO into 72 MO (=2191 dy), CMP Rev.11.</p>		
5	53-10-017-0392	<p>FORWARD FUSELAGE CIRCUMFERENTIAL SKIN SPLICES AND STRINGER SPLICES - INTERNAL Detailed Inspection of Circumferential Skin Splices and Stringer Splices - Region Below Str 17 - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD=72 MO; INTERVAL 72 MO (=2191 dy)</p>		
6	53-10-025-0930	<p>NOSE LANDING GEAR WHEELWELL CUTOUT STRUCTURE - INTERNAL General Visual Inspection of NLG Wheelwell Cutout Structure - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy)</p>		



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7	53-10-027-0384	<p>FORWARD FUSELAGE PASSENGER CABIN FLOOR STRUCTURE AND PASSENGER CABIN FLOOR TO FRAME ATTACHMENTS - INTERNAL General Visual Inspection of Passenger Floor Structure and Passenger Floor-to-Frame Attachment from FR 12 to aft of FR 20 Fuselage Junction between Fwd and Center Fuselage I - Internal Side of Passenger Cabin at Wet Areas.</p> <p>NOTE: Interval for: POST-MOD SB 190-53-0045 PART 1 OR POST-MOD SB 190-53-0077 = 72 MO (=2191 dy);</p> <p>NOTE: Stainless Steel strip removal is not required for aircraft Post-Mod SB 190-53-0077 configuration.</p>		
8	53-21-008-1108	<p>CENTER FUSELAGE I-TO-FORWARD FUSELAGE CIRCUMFERENTIAL SKIN SPLICES AND STRINGER SPLICES - INTERNAL Detailed Inspection of Center Fuselage I-to-Fwd Fuselage Circumferential Skin Splices and Stringer Splices Below Passenger Floor Level (below Stringer 17 LH/ RH, between Fr 20/23) - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).</p>		
9	53-21-009-0414	<p>FORWARD CARGO DOOR CUTOUT STRUCTURE AND LATCH BACKUP FITTINGS - INTERNAL General Visual Inspection of Fwd Cargo Door Cutout Structure and Latch Backup Fittings, from FR 24 to FR 27 and from STGR 17 RH to STGR 25 RH - Internal Side of Fuselage.</p> <p>NOTE: Interval changed from 66 MO into 72 MO (=2191 dy), CMP Rev.11.</p>		
10	53-21-014-1118	<p>CENTER FUSELAGE II-TO-CENTER FUSELAGE I CIRCUMFERENTIAL SKIN SPLICES AND STRINGER SPLICES - INTERNAL Detailed Inspection of Center Fuselage II-to-Center Fuselage I Circumferential Skin Splices and Stringer Splices Under Passenger Floor Level (below Stringer 17 LH/ RH, Between Fr 34/35) - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).</p>		
11	53-22-007-0961	<p>CENTER FUSELAGE II FORWARD PRESSURE BULKHEAD - INTERNAL General Visual Inspection of Fwd Pressure Bulkhead, Fwd Face, at Fr 45 - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).</p>		
12	53-22-008-0963	<p>CENTER FUSELAGE II AFT PRESSURE BULKHEAD - INTERNAL General Visual Inspection of Aft Pressure Bulkhead, Aft Face, at Fr 56 - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).</p>		



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13	53-22-012-0401	CENTER FUSELAGE II PASSENGER CABIN FLOOR STRUCTURE AND PASSENGER CABIN FLOOR-TO-FRAME ATTACHMENTS - INTERNAL General Visual Inspection of Passenger Cabin Floor Structure and Passenger Cabin Floor-to-Frame Attachments - Internal Side of Passenger Cabin. NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		
14	53-22-013-0402	MIDDLE AVIONICS COMPARTMENT ACCESS DOOR CUTOUT STRUCTURE - INTERNAL General Visual Inspection of Middle Avionics Compartment Access Door Cutout Structure, from Str 23 LH to Str 27 LH and from Fr 57 to Fr 58 - Internal Side of Fuselage. NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		
15	53-22-015-1168	CENTER FUSELAGE II CIRCUMFERENTIAL SKIN SPLICES AND STRINGER SPLICES - INTERNAL Detailed Inspection of Circumferential Skin Splices and Stringer Splices, at Fr 42/43 and at Fr 58/60 - Internal Side Fuselage. NOTE: The inspection of area behind PSU Supporting Structure, from STGR 5 to STGR 7, LH/RH, may be performed by using a borescope.		
16	53-23-001-1064	CENTER FUSELAGE III UPPER SKIN PANEL - INTERNAL General Visual Inspection of Upper Skin Panel - Internal Side of Dorsal Fin Fairing. NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		
17	53-23-005-0417	GVI - CENTER FUSELAGE III PASSENGER CABIN FLOOR STRUCTURES AND PASSENGER CABIN FLOOR TO FRAME ATTACHMENTS LOCATED AT WET AREAS - INTERNAL General Visual Inspection of Passenger Floor Structure and Passenger Floor to Frame Attachment, from FR 84 to FR 91 - Internal Side of Passenger Cabin at Wet areas. NOTE: Stainless Steel strip removal is not required for aircraft Post-Mod SB 190-53-0077 configuration. 3.50 NOTE: Interval is: PPOST-MOD SB 190-53-0045 PART 1 OR POST-MOD SB 190-53-0077 = 72 MO (=2191 dy); PRE-MOD SB 190-53-0045 PART 1 AND PRE-MOD SB190-53-0077= 48 MO (=1461 dy).		
18	53-23-007-0437	CENTER FUSELAGE III LONGITUDINAL SPLICES - INTERNAL Detailed Inspection of Longitudinal Splices Below Passenger Cabin (Str 23 RH and 25 LH), from Fr 73 to Fr 91 - Internal Side of Fuselage. NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		



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19	53-23-009-0419	GVI - AFT CARGO DOOR CUTOUT STRUCTURE AND LATCH BACKUP FITTINGS - INTERNAL General Visual Inspection of Aft Cargo Door Cutout Structure and Latch Backup Fittings, from FR 77 to FR 80 and from STGR 17 RH to STGR 26 RH - Internal Side of Fuselage. NOTE: Interval is: POST SB190-53-0067 = 72 MO (=2191 dy); PRE SB190-53-0067 = 48 MO (=1461 dy),		
20	53-23-018-0428	CENTER FUSELAGE III WASTE AND WATER ACCESS DOOR CUTOUT STRUCTURE - INTERNAL General Visual Inspection of Waste and Water Access Door Cutout Structure - Internal Side of Fuselage.		
21	53-30-005-1025	REAR FUSELAGE LOWER SKIN PANEL - INTERNAL General Visual Inspection of Rear Fuselage Lower Skin Panel from FR 91 to FR 100 and from STGR 21 LH to STGR 21 RH, including Frames, Stringers and Shearclips - Internal Side of Fuselage. NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		
22	53-30-011-1031	REAR FUSELAGE REAR PRESSURE BULKHEAD - INTERNAL General Visual Inspection of Rear Pressure Bulkhead Aft Face, at Non-Pressurized Side - Internal Side of Fuselage. NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		
23	53-30-020-1137	REAR FUSELAGE-TO-VERTICAL STABILIZER ATTACHMENT - INTERNAL Special Detailed Inspection of Rear Fuselage-to-Vertical Stabilizer Attachment, by using a Boroscope - Top Surface of Fuselage, Internal Side of Vertical Stabilizer. NOTE: TRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		
24	53-30-025-1142	REAR FUSELAGE REAR PRESSURE BULKHEAD RING - INTERNAL Detailed Inspection of Rear Fuselage Rear Pressure Bulkhead Ring, Lower Portion (below Str 17 LH/RH), at Non-Pressurized Side - Internal Side of Fuselage. NOTE: THRESHOLD=72 MO; INTERVAL=72 MO (=2191 dy).		
25	54-50-010-0470	PYLON FORWARD SIDE PANEL - INTERNAL General Visual Inspection of Pylon Side Panel, LH/RH - Internal Side - Lateral of Fairing and Skirts. NOTE: Interval is 72 MO (=2191 dy).		



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26	55-30-012-1013	<p>VERTICAL STABILIZER-TO-REAR FUSELAGE ATTACHMENT, ON THE VERTICAL STABILIZER - INTERNAL</p> <p>General Visual Inspection of Vertical Stabilizer-to- Rear Fuselage Attachment, on The Vertical Stabilizer - Internal Side of Stabilizer.</p> <p>NOTE: THRESHOLD = 2191 D , INTERVAL = 2191 D</p>		
27	57-01-002-1253	<p>GVI - WING STUB MAIN BOX LOWER SKIN PANEL UPPER SURFACE - INTERNAL</p> <p>General Visual Inspection of Wing Stub Main Box Lower Skin Upper Surface and Spllices, from Spar 1 to Spar 2 and from Wing Main Box Rib 1 LH to Rib 1 RH - Internal Side of Wing Stub Main Box.</p> <p>I: 108 MO (3287 Days) iaw CMP Rev 21.</p>		
28	57-01-004-1255	<p>GVI - WING STUB MAIN BOX RIBS - INTERNAL</p> <p>General Visual Inspection of Wing Stub Main Box Ribs from Spar 1 to Spar 2 and from Wing Main Box Rib 1 LH to Rib 1 RH - Internal Side of Wing Stub Main Box.</p> <p>I: 108 MO (3287 Days) iaw CMP Rev 21.</p>		
29	57-01-006-1260	<p>GVI - WING STUB SPAR 1 AFT SIDE - INTERNAL</p> <p>General Visual Inspection of Wing Stub Spar 1 Aft Side from Wing Main Box Rib 1 LH to Rib 1 RH - Internal Side of Wing Stub Main Box.</p> <p>I: 108 MO (3287 Days) iaw CMP Rev 21.</p>		
30	57-10-006-1310	<p>WING SPAR 2-TO-WING STUB TENSION JOINT AND BOLTS - EXTERNAL</p> <p>Detailed Inspection of Wing Spar 2 to Wing Stub Tension Joint and Bolts, LH/RH - External Side at MLG Compartment.</p> <p>NOTE: INTERVAL = 2191 Days.</p>		
31	57-50-003-1282	<p>WING SPAR 3 - EXTERNAL</p> <p>Detailed Inspection of Wing Spar 3 Fwd Face, from Rib 1A to Rib 5, LH/RH - External Side at Main Landing Gear Compartment.</p> <p>NOTE: Interval changed from "T: 72 MO I: 32 MO" To: "I: 72 MO" (=2191 dy), CMP Rev.11.</p>		
32	57-50-007-1248	<p>MAIN LANDING GEAR TRUNNION FITTINGS - EXTERNAL</p> <p>Detailed Inspection of Trunnion Fittings 2 and 3, between Rib 4 and Rib 5, LH/RH - External Side at Main Landing Gear Compartment.</p> <p>NOTE: Interval changed from "T: 72 MO I: 42 MO" To: "I: 72 MO" (=2191 dy), CMP Rev.11.</p>		

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33	57-50-008-0485	DET - MAIN LANDING GEAR ACTUATOR SUPPORT FITTING AND UPLOCK FITTINGS - EXTERNAL Detailed Inspection of Uplock Fittings, between Rib 1A and Rib 2, LH/RH - External Side at MLG Compartment. NOTE: Interval is 72 MO (=2191 dy).		
34	57-70-001-1290	SPOILER HINGE FITTINGS AND ACTUATOR FITTINGS -EXTERNAL General Visual Inspection of Spoiler Hinges Fittings and Spoiler Actuator Fittings, including Attachments at Spoiler Panel and WingSpars, LH/RH - External Side of Wing Fixed Trailing Edge. NOTE:Flaps have to be fully extended and Spoilers have to bedeployed. Interval: 7500 FC		