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SEQ	Task Card / EO	Task Card / EO Description	Certify	NOTE
1	25-11-01-002	<p>PILOT/CO-PILOT SEATS Detailed Inspection of Pilot/Co-Pilot Seats looking for possible degradation, cracks, damage, scratches and wear.</p> <p>NOTE: Interval changed from 6000 FH or 5000 FC Whichever occurs first, into 13750 fh, CMP Rev.11.</p>		
2	26-11-01-002	<p>ENGINE FIRE DETECTOR INTEGRITY AND ALARM SWITCH Operational Check of Engine Fire Detector Integrity and Alarm Switch. Remove the detector for a bench test.</p>		
3	26-12-01-002	<p>APU FIRE DETECTOR INTEGRITY AND ALARM SWITCH Operational Check of APU Fire Detector Integrity and Alarm Switch. Remove the detector to shop to perform the check.</p>		
4	27-11-01-001	<p>CONTROL YOKES CHAINS Lubrication of Control Yokes Chains.</p> <p>NOTE: Interval changed from 6000 fh into 15000 fh, CMP Rev. 11.</p>		
5	27-11-11-002C	<p>AILERON OVERRIDE UNIT Operational Check of Aileron Override Unit for Jamming / Disconnection.</p>		
6	27-41-01-005	<p>HORIZONTAL STABILIZER ACTUATOR NO-BACK MECHANISM Detailed Inspection of Horizontal Stabilizer Actuator No-Back Mechanism for broken or damaged components.</p> <p>NOTE: Interval changed from 9000 fh into 15000 fh, CMP Rev. 11.</p>		
7	27-83-01-003C	<p>OUTBOARD SLAT SKEW SENSOR Functional Check of Outboard Slat Skew Sensor.</p> <p>NOTE: ALI task, ONE STAR.</p> <p>NOTE 2: POST MOD SB190-27-0030.</p> <p>INTERVAL: 15 000 FH</p>		
8	52-31-02-002	<p>GVI - FORWARD CARGO DOOR DRAIN HOLES General Visual Inspection of FWD cargo door drain holes - Internal side of door for clogging.</p> <p>NOTE: Interval changed from 6000 fh into 15000 fh, CMP Rev.11</p>		



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9	53-10-014-0002	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Detailed Inspection of Upper Skin Panel around of Central Post and Upper Beam Attachment Holes - External Side of Fuselage. T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first		
10	53-10-014-0003	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Detailed Inspection of Upper Skin Panel around of Central Post and Lower Beam Attachment Holes - External Side of Fuselage. NOTE: Applicable for PRE-MOD SB 190-53-0041: T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first.		
11	53-10-014-0004	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Special Detailed Inspection (Eddy Current) of Upper Skin Panel around of Central Post and Lower Beam Attachment Holes to verify the Doubler located under the Skin Panel - External Side of Fuselage. T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first		
12	53-10-021-0006	NLG WHEEL WELL METALLIC STRUCTURE AT FRAME 08 ATTACHMENT - INTERNALL Detailed Inspection of NLG Wheel Well Metallic Bulkhead at FR 08 RH Attachment - Internal Side of RAT Compartment. T: 15225 FC I: 5755 FC		
13	53-10-024-0931	NOSE LANDING GEAR SHOCK STRUT AND UPPER DRAG BRACE HOUSING - EXTERNAL Detailed Inspection of NLG Shock Strut and Upper Drag Brace Housing - External Side of Structure. NOTE: Interval and Threshold changed from 6000 fc into 12000 fc, CMP Rev.11		
14	53-30-005-1024	REAR FUSELAGE LOWER SKIN PANEL - EXTERNAL General Visual Inspection of Lower Skin Panel, from Str 21 LH to Str 21 RH, and from Fuselage Junction between Fr 90 and Fr 91 to Fr 100 - External Side of Fuselage. NOTE: THRESHOLD=7000 FC; INTERVAL=7000 FC		



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15	53-30-008-0002	<p>REAR FUSELAGE ACCESS DOOR CUTOUT STRUCTURE - INTERNAL Detailed Inspection of Access Door Cutout Structure at Doubler to Skin Attachment Area, from FR 94 to FR 96 and from STR 13 RH to STR 27 RH - Internal Side of Fuselage.</p> <p>INTERVAL: 15746 FC</p>		
16	55-10-014-0987	<p>HORIZONTAL STABILIZER MIDDLE SPAR, FROM RIB 3L TO RIB 3R - INTERNAL Detailed Inspection of Middle Spar Fwd Face, from Rib 3 LH to Rib 3 RH - Internal Side of Rear Fuselage.</p> <p>NOTE: Interval is 3287 Days.</p>		
17	55-10-015-0989	<p>DET - HORIZONTAL STABILIZER REAR SPAR FROM RIB 3 TO TIP RIB (RIB 17) - INTERNAL Special Detailed Inspection (Borescope) of Rear Spar Fwd Face, from Rib 3 to Tip Rib (Rib 17), RH/LH - Internal Side of Horizontal Stabilizer.</p> <p>NOTE: Interval is 3165 Days.</p>		
18	55-10-021-0997	<p>HORIZONTAL STABILIZER HINGE BACKUP FITTINGS, ON THE HORIZONTAL STABILIZER RIB 3, LH/RH - INTERNAL Special Detailed Inspection (Borescope) of Hinge Backup Fittings, on Horizontal Stabilizer Rib 3 Inner Face, LH/RH - Internal Side of Horizontal Stabilizer at Rear Fuselage.</p> <p>NOTE: Interval is 3287 Days</p>		
19	55-20-002-1000	<p>ELEVATOR HINGES AND ACTUATORS FITTINGS - INTERNAL Detailed Inspection of Elevator Hinges and Actuators Fittings, LH/RH - Internal Side at Horizontal Stabilizer Rear Spar and Elevator area.</p> <p>NOTE: Elevators have to be removed</p> <p>NOTE: THRESHOLD = 16000 FC, INTERVAL = 13500 FC</p>		
20	55-30-008-1011	<p>VERTICAL STABILIZER REAR SPAR - INTERNAL Special Detailed Inspection of Rear Spar Aft Face from Root Rib to Tip Rib (Rib 14), by using a Boroscope - Internal Side Trailing Edge at Area Covered by Rudder.</p> <p>NOTE: THRESHOLD = 14667 FC , INTERVAL = 14667 FC</p>		
21	55-40-002-1002	<p>RUDDER HINGE AND ACTUATORS FITTINGS, ON THE VERTICAL STABILIZER AND RUDDER - INTERNAL Detailed Inspection of Rudder Hinge and Actuator Fittings - Area Covered by Rudder on Vertical Stabilizer and on Rudder Front Spar.</p> <p>NOTE: Rudder has to be removed.</p>		



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22	57-01-003-0490	<p>WING STUB REAR BOX UPPER SKIN AND STRUCTURES - EXTERNAL</p> <p>General Visual Inspection of External Side of Wing Stub Rear Box Upper Skin Panel Except Metallic Structure in the Fwd Outboard Corner, LH/RH - External Side at MLG Compartment.</p> <p>NOTE: THRESHOLD = 1642 D , INTERVAL = 1642 D.</p> <p>NOTE: Interval changed from 54 MO into 72 MO, CMP Rev.11</p>		
23	57-10-004-0002	<p>DET - WING SPAR 2 - INTERNAL</p> <p>Detailed Inspection of Wing Spar 02 Forward Face, from Rib 10 to Rib 13, Including its attachments to Wing Main Box Lower Skin, LH/RH - Inside of Wing Main Box.</p> <p>NOTE: T: 15010 FC I: 15010 FC OR T: 19963 FH I: 19963 FH Whichever occurs first</p>		
24	57-50-004-1223	<p>WING SPAR 3 TO WING STUB ATTACHMENTS - EXTERNAL</p> <p>Detailed Inspection of Wing Spar 3 to Wing Stub Attachments including Upper Fitting, Tie Rod and Integral Fitting of Spar 3, LH/RH - External Side at MLG Compartment.</p> <p>NOTE: THRESHOLD - 2191 D , INTERVAL - 1521 D.</p>		
25	57-50-005-0002	<p>WING FIXED TRAILING EDGE RIB 9A - INTERNAL</p> <p>Detailed Inspection of Wing Fixed Trailing Edge Rib 9A Upper Flange, LH/RH - Inside Wing Trailing Edge Box.</p> <p>INTERVAL: 15225 FC OR 20249FH</p> <p>NOTE: Task applicable for PRE-MOD SB 57-0028</p>		
26	57-50-029-1299	<p>AFT FLAP ROLLER FITTING AT TRACK 1</p> <p>Detailed Inspection of Aft Flap Roller Fitting at Track 1, LH/RH - External Area.</p> <p>NOTE: Flaps have to be fully extended.</p> <p>NOTE: THRESHOLD - 8000 FC , INTERVAL - 8000 FC.</p>		
27	EO-AOB-25-27	<p>ARM-DISARM DOOR ASSEMBLY-HINGE, FWD & AFT PASSENGER DOOR - REPLACEMENT</p> <p>Perform replacement of hinge on Arm-Disarm door assembly - FWD & AFT Passenger door with new reinforced hinge iaw AMM 25-23-22-000/400-801-A.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book.</p>		



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28	EO-AOB-27-14	<p>FLIGHT CONTROLS - DISCONNECT HANDLE - PROTECTOR PLATE INSTALLATION</p> <p>Perform modification of Aileron/Elevator Disconnect Handle iaw SB 190-27-0040 3. Accomplishment Instructions - latest revision.</p> <p>NOTE for Planning: After accomplishment of this EO, make a record in a/c log book, modification log book, SB status list and inform Embraer through to Reliability Report.</p>		
29	EO-AOB-32-32	<p>REPLACEMENT OF THE SUPPORT FLANGE OF THE SWIVEL JOINTS</p> <p>1. Prepare the aircraft iaw AMM 32-31-05-000-801-A.</p> <p>2. a) LH MLG: - Disconnect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-03A item 320) iaw AMM 20-10-03-000/400-801-A. - Replace BONDING SUPPORT PN: 190-70161-001 with PN: 190-70197-001 (AIPC 32-11-05-03A item 380A) iaw AMM 20-10-01-910-801-A. - Connect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-03A item 320A) iaw AMM 20-10-03-000/400-801-A and SNL 190-32-0037.</p> <p>2. b) RH MLG: - Disconnect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-05A item 320) iaw AMM 20-10-03-000/400-801-A. - Replace BONDING SUPPORT PN: 190-70161-001 with PN: 190-70197-001 (AIPC 32-11-05-05A item 380A) iaw AMM 20-10-01-910-801-A. - Connect the HYDRAULIC SWIVEL JOINT ASSEMBLY (AIPC 32-11-05-05A item 320A) iaw AMM 20-10-03-000/400-801-A and SNL 190-32-0037.</p> <p>3. Close the job iaw AMM 32-31-05-400-801-A.</p> <p>Note: Lifting the aircraft on jacks is required. Bleeding of the hydraulic system No. 2 is required. An operational test of the landing gear is required.</p>		



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30	EO-AOB-52-13	<p>FORWARD PASSENGER DOOR - REPLACEMENT OF THE BONDING STRAPS AND REPLACEMENT/PROTECTION OF THE SENSOR HARNESS</p> <p>Perform replacement of the bonding straps and protection of the sensor harness iaw PART I of SB 190-52-0046, following next steps:</p> <ol style="list-style-type: none"> 1. Perform replacement of four flat-section bonding straps by two new cylindrical section bonding straps. 2. Perform installation of anti-rotation device and a protection sheet. 3. Install a protection on the sensor harness from the hinge to door and from the hinge to fuselage paths. 4. Rework the door linings iaw C&D Zodiac SB 170-18616-25-386. 5. Reidentify the FWD passenger door iaw Figure 11 of SB 190-52-0046. <p>Note 1: Metal sheet mechanic is needed.</p> <p>Note 2: After accomplishment of SB 190-52-0046 do not forget to make aircraft log book entry.</p>		
31	EO-AOB-53-47	<p>FUSELAGE - FORWARD ELECTRONIC RACK - STRUCTURAL PARTS REPLACEMENT AND REINFORCEMENTS INSTALLATION</p> <p>Perform modification iaw SB190-53-0074 latest revision, 3.Accomplishment instructions.</p> <p>NOTE: p/n: 170-10882-401 support, after reworked need to be reidentified with new p/n: 170-10882-603.</p> <p>NOTE for planning: The due date is a flexible.</p> <p>NOTE for planning: Sheet metal mechanic is required.</p> <p>NOTE for engineering: After accomplishment of this EO, make accomplishment note in:</p> <ul style="list-style-type: none"> - SB status; - Modification log book; - and inform Embraer that SB has been incorporated. 		
32	EO-AOB-53-50	<p>FUSELAGE - REINFORCEMENT OF FRAMES OF THE FWD FUSELAGE CLOSE TO NLG BAY-MODIFICATION.</p> <p>Perform modification of Reinforcement of frames of the FWD fuselage close to NLG bay iaw SB 190-53-0026-3. Accomplishment Instructions.</p> <p>NOTE: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After performed of this EO, make a record in Modification log book, SB status list.</p>		



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33	EO-AOB-53-51	<p>FUSELAGE-STRUCTURAL REINFORCEMENT OF SKIN PANEL IN THE WINDSHIELD CUTOUT STRUCTURE - MODIFICATION. Perform modification of Skin Panel in the Windshield Cutout Structure iaw SB 190-53-0041R01-3.Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification lod book and SB status list.</p>		
34	EO-AOB-53-52	<p>FORWARD FUSELAGE - REPLACEMENT OF FASTENER IN FRAME 7 - MODIFICATION. Perform modification of fastener in Frame 7 iaw SB 190-53-0043 R02-3. Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
35	EO-AOB-53-53	<p>FORWARD FUSELAGE - REINFORCEMENT OF NLG BULKHEAD WEB AND FRAME 8 - MODIFICATION. Perform modification of reinforcement of NLG Bulkhead web and frame 8 iaw SB 190-53-0049 R02-3. Accomplishment Instructions, SB 190-53-0049 IN1 and SB 190-53-0049 IN2.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
36	EO-AOB-53-54	<p>FORWARD FUSELAGE - LATERAL POST- T SHAPE REPLACEMENT- MODIFICATION. Perform modification of reinforcement of T shape lateral post iaw SB 190-53-0050 R01-3. Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
37	EO-AOB-53-55	<p>FUSELAGE - STRUCTURAL REINFORCEMENT OF LOWER BEAM IN THE COCKPIT WINDSHIELD STRUCTURE- MODIFICATION. Perform modification of Structural Reinforcement of Lower Beam in the Cockpit Windshield Structure iaw SB 190-53-0053-3. Accomplishment Instructions and SB 190-53-0053 IN1.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		



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38	EO-AOB-53-57	<p>FORWARD FUSELAGE - REINFORCEMENT OF DIRECT VISION WINDOWS- MODIFICATION. Perform modification of reinforcement of T shape lateral post iaw SB 190-53-0052-3. Accomplishment Instructions, SB 190-53-0052 IN 1 and SB 190-53-0052 IN 2.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p> <p>NOTE for CAMO: Recommended to be performed before TCN: 20000 FC.</p>		
39	EO-AOB-53-58	<p>FORWARD FUSELAGE - FORWARD PASSENGER AND SERVICE DOOR CUTOUT STRUCTURE REINFORCEMENT - MODIFICATION. Perform modification of reinforcement of Forward Passenger and Service door cutout structure iaw SB 190-53-0059 R01. 3.Accomplishment Instructions PART I and PART II.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p> <p>NOTE for CAMO: Recommended to be performed before TCN: 20000 FC.</p>		
40	EO-AOB-53-59	<p>FUSELAGE-WING-TO-FUSELAGE FAIRING AFT PANEL-FLAP MECHANISM ACCESS DOOR - ELECTRICAL BONDING IMPROVEMENT - MODIFICATION. Perform modification of Wing-to-Fuselage fairing AFT panel-Flap mechanism access door iaw SB 190-53-0091 3. Accomplishment Instructions, SB 190-53-0091 IN1, SB 190-53-0091 IN2 and SB 190-53-0091 IN3.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB Status list.</p>		
41	EO-AOB-53-67	<p>FWD PASSENGER DOOR FRAME FITTING - SEAL DEPRESSOR - PERMANENT REPAIR. Perform permanenet repair of FWD Passenger door frame fitting - seal depressor iaw SRM 53-10-25/Repair 51P.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for MCC: After accomplishment of this EO, make a record in ADAL. Ref. to ADAL item 16.</p> <p>NOTE: All sheet metal works performed by _____</p>		



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42	EO-AOB-55-08	<p>VERTICAL STABILIZER STRUCTURAL REINFORCEMENT - NEW RIB INSTALLATION Perform modification on the Vertical Stabilizer by installing a new Rib iaw SB 190-55-0003.</p> <p>NOTE 1: For Planning: NDT and Sheet mechanics are needed for this modification.</p> <p>NOTE 2: For Planning: This modification requires rudder removing and installing. For that reason, modification should be planned during accomplishment of the CMP tasks: 55-30-008-0004, 55-40-002-0001 and/or 55-40-002-1002.</p> <p>NOTE 3: For Engineering: After accomplishment, make note in Aircraft Log Book: - "SB 190-55-0003 is performed by Montenegro Airlines, @...."</p>		
43	EO-AOB-55-17	<p>INSTALLATION OF DIMPLED WASHER IN VERTICAL STABILIZER AERODYNAMIC SEALS ATTACHING PARTS - MODIFICATION Perform installation of dimpled washer in vertical stabilizer aerodynamic seals attaching parts iaw SB190-55-0006, 3.Accomplishment Instructions.</p> <p>NOTE: Aircraft is applicable to GROUP 2.</p> <p>NOTE: Perform rework the existing vertical stabilizer aerodynamic seals.</p> <p>NOTE: Perform installation of missing aerodynamic seal on No.1 LH postion (p/n: 170-63410-403).</p> <p>NOTE: After completion update ADAL item 125.</p> <p>NOTE for planning: Sheet metal mechanic is required.</p> <p>NOTE for engineering: After accomplishment of this EO, make accomplishment note in: - SB status; - Modification log book; - and inform Embraer that SB has been incorporated.</p>		
44	EO-AOB-55-18	<p>STABILIZER - VERTICAL STABILIZER TRAILING EDGE RIBS REINFORCEMENT - MODIFICATION Perform modification of Vertical Stabilizer Trailing Edge Ribs iaw SB 190-55-0008 and SB190-55-0008 IN01.</p> <p>NOTE: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification Log book and SB status list.</p>		



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45	EO-AOB-55-19	<p>STABILIZER - ELEVATOR ROOT FAIRING TRAILING EDGE - ADDITIONAL DRAIN HOLE - MODIFICATION Perform modification of Elevator Root Fairing Trailing Edge by drilling additional hole iaw SB 190-55-0007.</p> <p>NOTE: After accomplishment of modification perform reidentify the elevator (LH/RH) by crossing out existing part number (P/N 190-62062-411) and marking the new part number next to the crossed out one according to the 2.E. REIDENTIFIED PARTS of this SB.</p> <p>NOTE for Engineering: After accomplishment of this SB, make a record in Modification log book, SB status list.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p>		
46	EO-AOB-55-21	<p>RUDDER AND RUDDER TIP FAIRING, RH & LH SIDE -PERMANENT REPAIR Perform permanent repair on surface protection of the Rudder and Rudder Tip Fairing, RH & LH side iaw SRM PART II 55-40-00/REPAIR 53P.</p> <p>NOTE: Composite repair will be done in out station MRO.</p> <p>NOTE for MCC: After accomplishment of this EO, make a record in ADAL item 142.</p> <p>NOTE for Planning: After accomplishment of this EO, terminate EO-AOB-55-23.</p> <p>NOTE: This EO is revised with ETD2018-190-006493-A as of 22.05.2018. Tepmorary allowable damages are valid for the next 1300 FC as of 17. Jan 2018, until the a/c reach CSN: 13422 FC.</p>		
47	EO-AOB-57-19	<p>WINGS - STRUCTURAL REINFORCEMENT OF THE FIXED TRAILING EDGE RIB 9A-MODIFICATION. Perform modification of Structural Reinforcement of the Fixed Trailing Edge Rib 9A iaw SB 190-57-0028-3. Accomplishment Instructions PART I nad PART II and SB 190-57-0028 IN 3.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
48	EO-AOB-57-20	<p>WING-STRUCTURAL REINFORCEMENT OF SPAR II BTW RIBS 10 AND 12-MODIFICATION. Perform modification of LH nad RH wing- SPAR II btw RIBS 10 and 12 iaw SB190-57-0044R01 PART I & PART II and SB190-57-0044R01 IN1.</p> <p>NOTE for Planning: Sheet metal specilaist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		



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49	EO-AOB-57-21	<p>WINGS - STRUCTURAL REINFORCEMENT OF WING STUB SIDE FITTING - MODIFICATION. Perform modification of reinforcement of wing stub side fitting iaw SB 190-57-0046. 3.Accomplishment Instructions.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p> <p>NOTE for CAMO: Recommended to be performed before TCN: 20000 FC.</p>		
50	EO-AOB-57-22	<p>WINGS - REINFORCEMENT OF FIXED TRAILING EDGE RIB 5 AND TRUNNION - MODIFICATION. Perform modification of reinforcement of fixed trailing edge rib 5 and trunnion iaw SB 190-57-0051 . 3.Accomplishment Instructions PART I and PART II.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p> <p>NOTE for CAMO: Recommended to be performed before TCN: 20000 FC.</p>		
51	Z250-002	<p>GVI - CENTER FUSELAGE III, AREA COVERED BY GALLEY AND TOILET, AND AREA BEHIND GALLEY AND TOILET TO REAR PRESSURE BULKHEAD FWD FACE - INTERNAL General Visual Inspection of Center Fuselage III, area covered by Galley and Toilet, area behind Galley and Toilet to Rear Pressure Bulkhead FWD Face and area below Floor Panels - Internal Side of Fuselage with Floor Panels Removal.</p> <p>NOTE: EWIS. Galley and Toilet have to be removed.</p> <p>NOTE: Interval is 15000 fh.</p>		