



WORKPACK ACCOUNTABILITY FORM

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SEQ	Task Card / EO	Task Card / EO Description	Certify	NOTE
1	26-11-01-002	ENGINE FIRE DETECTOR INTEGRITY AND ALARM SWITCH Operational Check of Engine Fire Detector Integrity and Alarm Switch. Remove the detector for a bench test.		
2	26-12-01-002	APU FIRE DETECTOR INTEGRITY AND ALARM SWITCH Operational Check of APU Fire Detector Integrity and Alarm Switch. Remove the detector to shop to perform the check.		
3	27-11-11-002C	AILERON OVERRIDE UNIT Operational Check of Aileron Override Unit for Jamming / Disconnection.		
4	27-83-01-003C	OUTBOARD SLAT SKEW SENSOR Functional Check of Outboard Slat Skew Sensor. NOTE: ALI task, ONE STAR. NOTE 2: POST MOD SB190-27-0030. INTERVAL: 15 000 FH		
5	29-11-00-001C	FNC - NO.1 & 2 HYDRAULIC SYSTEM BOOTSTRAP SYSTEMS Functional Check of No.1 & 2 Hydraulic System Bootstrap Systems. Check proper operation of bootstrap main components (Priority Valve and other main components) through ACMPs operation and pressure monitoring. NOTE: Interval is 22500 fh. NOTE: CMR one star.		
6	29-11-01-001	EDP DEPRESSURIZATION SOLENOID VALVE Operational Check of EDP Depressurization Solenoid Valve.		
7	29-33-02-002C 198	FNC - NO.1 HYDRAULIC SYSTEM TEMPERATURE SWITCHES Functional Check of No.1 Hydraulic System Temperature Switches. To be complied with replacement serviceable parts. NOTE: Interval is 22500 fh. NOTE: CMR one star.		
8	29-33-02-002C 199	FNC - NO.2 HYDRAULIC SYSTEM TEMPERATURE SWITCHES Functional Check of No.2 Hydraulic System Temperature Switches. To be complied with replacement serviceable parts. NOTE: Interval is 22500 fh. NOTE: CMR one star.		



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9	30-21-00-001	ENGINE THERMAL ANTI-ICING SYSTEM Operational Check of Engine Thermal Anti-Icing System in Manual Mode. NOTE: Interval changed from 20000 fh into 22500 fh, CMP Rev.11.		
10	52-10-003-0003	FORWARD PASSENGER DOOR STOPS AT FUSELAGE - GVI General Visual Inspection of Fwd Passenger Door-Stops and Surrounding Structures at Fuselage Frame - External Side of Fuselage THRESHOLD: 17299 FC, INTERVAL: 5102 FC		
11	52-10-003-0006	AFT PASSENGER DOOR STOPS AT FUSELAGE - EXTERNAL General Visual Inspection of Aft Passenger Door-Stops and surrounding Structures at Fuselage Frame - External Side of Fuselage. THRESHOLD: 17299 FC, INTERVAL: 5102 FC		
12	52-40-003-0003	FORWARD SERVICE DOOR STOPS AT FUSELAGE - EXTERNAL General Visual Inspection of Fwd Service Door-Stops and surrounding Structures at Fuselage - External Side of Fuselage. THRESHOLD: 17299 FC INTERVAL: 5102 FC		
13	52-40-003-0006	AFT SERVICE DOOR STOPS AT FUSELAGE - EXTERNAL General Visual Inspection of Aft Service Door-Stops and surrounding Structures at Fuselage - External Side of Fuselage THRESHOLD: 17299 FC, INTERVAL: 5102 FC		
14	53-10-008-0926	FORWARD PASSENGER DOOR CUTOUT STRUCTURE - INTERNAL General Visual Inspection of Fwd Passenger Door Cutout Structure, from Fr 15 to Fr 19 A, Area Covered by Scuff Plate. NOTE: TRESHOLD=15333 FC; INTERVAL 15333 FC.		
15	53-10-009-0934	FORWARD SERVICE DOOR CUTOUT STRUCTURE - INTERNAL General Visual Inspection of Fwd Service Door Cutout Structure,from Fr 13 to Fr 18,Area Covered by Scuff Plate. NOTE: TRESHOLD=15333 FC; INTERVAL=15333 FC		



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16	53-10-014-0002	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Detailed Inspection of Upper Skin Panel around of Central Post and Upper Beam Attachment Holes - External Side of Fuselage. T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first		
17	53-10-014-0003	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Detailed Inspection of Upper Skin Panel around of Central Post and Lower Beam Attachment Holes - External Side of Fuselage. NOTE: Applicable for PRE-MOD SB 190-53-0041: T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first.		
18	53-10-014-0004	FORWARD FUSELAGE UPPER SKIN PANEL - EXTERNAL Special Detailed Inspection (Eddy Current) of Upper Skin Panel around of Central Post and Lower Beam Attachment Holes to verify the Doubler located under the Skin Panel - External Side of Fuselage. T: 15144 FC I: 3096 FC OR T: 20141 FH I: 4117 FH Whichever occurs first		
19	53-10-021-0006	NLG WHEEL WELL METALLIC STRUCTURE AT FRAME 08 ATTACHMENT - INTERNALL Detailed Inspection of NLG Wheel Well Metallic Bulkhead at FR 08 RH Attachment - Internal Side of RAT Compartment. T: 15225 FC I: 5755 FC		
20	53-10-027-0384	GVI - FORWARD FUSELAGE PASSENGER CABIN FLOOR STRUCTURE AND PASSENGER CABIN FLOOR-TO-FRAME ATTACHMENTS - INTERNAL General Visual Inspection of Passenger Floor Structure and Passenger Floor-to-Frame Attachment from FR 12 to aft of FR 20 Fuselage Junction between Fwd and Center Fuselage I - Internal Side of Passenger Cabin at Wet Areas. NOTE: Interval for: POST SB190-53-0045 PART 1 = 72 MO (=2191 dy); PRE SB190-53-0045 PART 1 = 48 MO (=1461 dy).		



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21	53-21-005-0411	<p>CENTER FUSELAGE I PASSENGER CABIN FLOOR STRUCTURE AND PASSENGER CABIN FLOOR-TO-FRAME ATTACHMENTS LOCATED AT DRY AREAS - INTERNAL</p> <p>General Visual Inspection of Passenger Cabin Floor Structure and Passenger Cabin Floor-to- Frame Attachments Located at Dry Areas (floor structures located out of galley, toilet and door areas) - Internal Side of Passenger Cabin at Floor Structures Dry Areas.</p> <p>TRESHOLD 2922 DY , INTERVAL 2677 DY</p>		
22	53-22-005-0958	<p>CENTER FUSELAGE II PASSENGER WINDOW CUTOUT STRUCTURE (SKIN AND WINDOW FRAME) - INTERNAL</p> <p>General Visual Inspection of Passenger Window Cutout Structure (Skin and Window Frame), from Str 08 L/R to Str 11 L/R and from Fr34 to Fr74 -Internal Side of Fuselage</p> <p>TRESHOLD: 16667FC, INTERVAL: 16667 FC</p>		
23	53-23-005-0417	<p>GVI - CENTER FUSELAGE III PASSENGER CABIN FLOOR STRUCTURES AND PASSENGER CABIN FLOOR-TO-FRAME ATTACHMENTS LOCATED AT WET AREAS - INTERNAL</p> <p>General Visual Inspection of Passenger Cabin Floor Structures and Passenger Cabin Floor-to-Frame Attachments Located at Wet Areas - Floor Structures Located Under Galley/Toilet and in front of Door Areas (FR 84 to FR 91) - Internal Side of Fuselage.</p> <p>NOTE: Interval is: POST-MOD SB 190-53-0045 PART 1 = 72 MO (=2191 dy); PRE-MOD SB 190-53-0045 PART 1 = 48 MO (=1461 dy).</p>		
24	53-23-009-0419	<p>GVI - AFT CARGO DOOR CUTOUT STRUCTURE AND LATCH BACKUP FITTINGS - INTERNAL</p> <p>General Visual Inspection of Aft Cargo Door Cutout Structures and Latch Backup Fittings, from FR 77 to FR 80 and from STGR 17 RH to STGR 26 RH - Internal Side of Fuselage</p> <p>NOTE: Interval is: POST SB190-53-0067 = 72 MO (=2191 dy); PRE SB190-53-0067 = 48 MO (=1461 dy), CMP rev 11.</p>		
25	53-23-014-0427	<p>AFT PASSENGER DOOR CUTOUT STRUCTURE - INTERNAL</p> <p>General Visual Inspection of Aft Passenger Door Cutout Structure, from Fr 84 to Fr 88, below Floor Level (below Str 17) - Internal Side of Fuselage.</p> <p>NOTE1: CIC reapplication is required NOTE2: TRESHOLD=72 MO; INTERVAL=48 MO</p>		



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26	53-23-018-0428	<p>CENTER FUSELAGE III WASTE AND WATER ACCESS DOOR CUTOUT STRUCTURE - INTERNAL General Visual Inspection of Waste and Water Access Door Cutout Structure - Internal Side of Fuselage.</p> <p>NOTE: POST-MOD SB 190-53-0056 interval = 72 MO (=2191 dy); PRE-MOD SB 190-53-0056 interval = 48 MO (=1461 dy).</p> <p>NOTE: Threshold changed from 72MO into 48MO, Interval changed from 64MO into 48MO iaw MRB rev 07 dated May 08/14 and CMP Rev 11. See MRB Introduction paragraph 6.H.</p>		
27	53-30-008-0002	<p>REAR FUSELAGE ACCESS DOOR CUTOUT STRUCTURE - INTERNAL Detailed Inspection of Access Door Cutout Structure at Doubler to Skin Attachment Area, from FR 94 to FR 96 and from STR 13 RH to STR 27 RH - Internal Side of Fuselage.</p> <p>INTERVAL: 15746 FC</p>		
28	53-30-018-1132	<p>HORIZONTAL STABILIZER ACTUATOR FITTINGS, BACKUP FITTINGS AND ATTACHMENTS - INTERNAL Detailed Inspection of Horizontal Stabilizer Actuator Fittings, Backup Fittings and their attachments to FR 96 - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD=16000 FC; INTERVAL=16000 FC</p>		
29	55-10-013-0984	<p>HORIZONTAL STABILIZER FRONT SPAR - INTERNAL Detailed Inspection of Front Spar Aft Face, LH/RH - Internal Side of Stabilizer.</p> <p>NOTE: THRESHOLD: 17333 FC INTERVAL: 17333 FC</p>		
30	55-10-014-0987	<p>HORIZONTAL STABILIZER MIDDLE SPAR, FROM RIB 3L TO RIB 3R - INTERNAL Detailed Inspection of Middle Spar Fwd Face, from Rib 3 LH to Rib 3 RH - Internal Side of Rear Fuselage.</p> <p>NOTE: THRESHOLD: 16000 FC INTERVAL: 16000 FC</p>		
31	55-10-015-0989	<p>DET - HORIZONTAL STABILIZER REAR SPAR FROM RIB 3 TO TIP RIB (RIB 17) - INTERNAL Detailed Inspection of Rear Spar Fwd Face, from Rib 3 to Tip Rib (Rib 17), RH/LH - Internal Side of Stabilizer.</p> <p>NOTE: Elevators have to be removed. NOTE: THRESHOLD: 16000 FC INTERVAL: 16000 FC</p>		



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32	55-10-021-0997	<p>HORIZONTAL STABILIZER HINGE BACKUP FITTINGS, ON THE HORIZONTAL STABILIZER RIB 3, LH/RH - INTERNAL Detailed Inspection of Hinge Backup Fittings, on Horizontal Stabilizer Rib 3 Fwd Face, LH/RH - Internal Side of Stabilizer at Rear Fuselage.</p> <p>NOTE: THRESHOLD = 16000 FC, INTERVAL = 16000 FC</p>		
33	55-20-002-1000	<p>ELEVATOR HINGE AND ACTUATOR FITTINGS - INTERNAL Detailed Inspection of Elevator Hinge and Actuator Fittings, LH/RH - Internal Side at Rear Spar and Elevator Area.</p> <p>NOTE: THRESHOLD = 16000 FC, INTERVAL = 16000 FC</p>		
34	55-40-002-1002	<p>DET - RUDDER HINGE AND ACTUATOR FITTINGS - INTERNAL Detailed Inspection of Rudder Hinge and Actuator Fittings - Area Covered by Rudder. NOTE: Rudder has to be removed. NOTE: THRESHOLD = 16000 FC , INTERVAL = 16000 FC.</p>		
35	57-01-003-0490	<p>WING STUB REAR BOX UPPER SKIN AND STRUCTURES - EXTERNAL General Visual Inspection of External Side of Wing Stub Rear Box Upper Skin Panel Except Metallic Structure in the Fwd Outboard Corner, LH/RH - External Side at MLG Compartment.</p> <p>NOTE: THRESHOLD = 1642 D , INTERVAL = 1642 D.</p> <p>NOTE: Interval changed from 54 MO into 72 MO, CMP Rev.11</p>		
36	57-01-003-1311	<p>WING STUB REAR BOX UPPER SKIN AND STRUCTURES – INTERNAL General Visual Inspection of Wing Stub Rear Box Floor Beam Structure (from Fr 52 to Fr 56) - Internal Side of Fuselage.</p> <p>TRESHOLD: 16667 FC, INTERVAL: 16667 FC</p>		
37	57-01-005-1257	<p>WING STUB REAR BOX RIBS AND INTERCOSTALS - EXTERNAL General Visual Inspection of Wing Stub Rear Box Ribs 1 and 2 (RH/LH) and Intercostals from Spar 2 to Spar 3 - External Side at MLG Compartment.</p> <p>NOTE: Interval changed from 6000 FC into 72 MO (=2191 dy), CMP Rev.11.</p>		
38	57-01-006-1261	<p>WING STUB SPAR 1 - INTERNAL Detailed Inspection of Wing Stub Spar 1, from Wing Main Box Rib 1 LH to Rib 1 RH - Internal Side of Fuselage.</p> <p>NOTE: THRESHOLD = 16667 FC , INTERVAL = 16667 FC.</p>		



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39	57-01-007-1262	<p>WING STUB SPAR 2 - INTERNAL Detailed Inspection of Wing Stub Spar 2, from Wing Main Box Rib 1 LH to Rib 1 RH - Internal Side of Wing Stub Box.</p> <p>NOTE: THRESHOLD = 15333 FC , INTERVAL = 15333 FC.</p>		
40	57-10-004-0002	<p>DET - WING SPAR 2 - INTERNAL Detailed Inspection of Wing Spar 02 Forward Face, from Rib 10 to Rib 13, Including its attachments to Wing Main Box Lower Skin, LH/RH - Inside of Wing Main Box.</p> <p>NOTE: T: 15010 FC I: 15010 FC OR T: 19963 FH I: 19963 FH Whichever occurs first</p>		
41	57-10-006-1310	<p>WING SPAR 2-TO-WING STUB TENSION JOINT AND BOLTS - EXTERNAL Detailed Inspection of Wing Spar 2-to-Wing Stub Joint and Bolts, LH/RH - External Side at MLG Compartment.</p> <p>NOTE: THRESHOLD = 8000 FC , INTERVAL = 8000 FC.</p>		
42	57-50-002-1218	<p>GVI - FIXED TRAILING EDGE LOWER SKIN General Visual Inspection of Fixed Trailing Edge Lower Skin Panel at Area Covered by Pylon and Flap Fairing, LH/RH.</p> <p>NOTE: THRESHOLD - 16667 FC , INTERVAL - 16667 FC.</p>		
43	57-50-004-1223	<p>WING SPAR 3 TO WING STUB ATTACHMENTS - EXTERNAL Detailed Inspection of Wing Spar 3 to Wing Stub Attachments including Upper Fitting, Tie Rod and Integral Fitting of Spar 3, LH/RH - External Side at MLG Compartment.</p> <p>NOTE: THRESHOLD - 2191 D , INTERVAL - 1521 D.</p>		
44	57-50-005-0002	<p>WING FIXED TRAILING EDGE RIB 9A - INTERNAL Detailed Inspection of Wing Fixed Trailing Edge Rib 9A Upper Flange, LH/RH - Inside Wing Trailing Edge Box.</p> <p>INTERVAL: 15225 FC OR 20249FH</p> <p>NOTE: Task applicable for PRE-MOD SB 57-0028</p>		
45	57-50-029-1299	<p>AFT FLAP ROLLER FITTING AT TRACK 1 Detailed Inspection of Aft Flap Roller Fitting at Track 1, LH/RH - External Area.</p> <p>NOTE: Flaps have to be fully extended.</p> <p>NOTE: THRESHOLD - 8000 FC , INTERVAL - 8000 FC.</p>		



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46	EO-AOA-51-01	<p>POLYURETHANE FILM - NEW INSTALLATION</p> <p>1) Install polyurethane film on the aft nose landing gear doors iaw SB 190-51-002, Part I.</p> <p>2) Replace the polyurethane film of the LH and RH winglet with new ones iaw SB 190-51-002, Part II.</p> <p>3) Replace the polyurethane film of the vertical stabilizer tip fairing with new ones iaw SB 190-51-0002, Part III.</p> <p>4) Install polyurethane film on the LH and RH wing tip iaw SB 190-51-0002, Part IV.</p> <p>NOTE (for engineering): After accomplishment of SB 190-51-0002 do not forget to make aircraft log book entry.</p>		
47	EO-AOA-53-36	<p>FORWARD CARGO DOOR - CORROSION IN CARGO DOOR SEAL REST</p> <p>Perform SB 190-53-0067 Item 3. ACCOMPLISHMENT INSTRUCTIONS - B. PART I - MODIFICATION IN THE FORWARD CARGO DOOR FRAME</p> <p>NOTE for engineering: After accomplishment of this modification make note in aircraft log book: `SB190-53-0067 performed by, @.....fh, fc, (date).`</p>		
48	EO-AOA-53-37	<p>AFT CARGO DOOR - CORROSION IN CARGO DOOR SEAL REST</p> <p>Perform SB 190-53-0067 Item 3. ACCOMPLISHMENT INSTRUCTIONS - C. PART II - MODIFICATION IN THE AFT CARGO DOOR FRAME</p> <p>NOTE for engineering: After accomplishment of this modification make note in aircraft log book: `SB190-53-0067 performed by, @.....fh, fc, (date).`</p>		
49	EO-AOA-53-38	<p>RESTORATION OF MINOR PAINT DAMAGE ON WATER SERVICE PANEL (ADAL ITEM 100)</p> <p>1. Perform peeling of paint as much as necessary using an aluminium-oxyde grain No. 320 or finer to sand surface if applicable.</p> <p>2. Perform surface preparation as basis for painting iaw SRM 51-23-04.</p> <p>3. Perform restoration of paint damage on ater service panel (ADAL Item 100) iaw SRM 51-23-06.</p> <p>NOTE: After restoration is performed close ADAL Item 100.</p>		



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50	EO-AOA-55-09	<p>VERTICAL STABILIZER STRUCTURAL REINFORCEMENT - NEW RIB INSTALLATION</p> <p>Perform modification on the Vertical Stabilizer by installing a new Rib iaw SB 190-55-0003.</p> <p>NOTE 1: For Planning: NDT and Sheet mechanics are needed for this modification.</p> <p>NOTE 2: For Planning: This modification requires rudder removing and installing. For that reason, modification should be planned during accomplishment of the CMP task 55-40-002-1002.</p> <p>NOTE 3: For Engineering: After accomplishment, make note in Aircraft Log Book: - "SB 190-55-0003 is performed by Montenegro Airlines, @...."</p>		
51	EO-AOA-55-12	<p>RESTORATION OF MINOR PAINT DAMAGE ON LH ELEVATOR BOTTOM SIDE (ADAL ITEM 101).</p> <p>1. Perform peeling of paint as much as necessary using an aluminium-oxyde grain No. 320 or finer to sand surface if applicable.</p> <p>2. Perform surface preparation as basiss for painting iaw SRM 51-23-04.</p> <p>3. Perform restoration of paint damage on LH Elevator bottom side (ADAL Item 101) iaw SRM 51-23-06.</p> <p>NOTE: After restoration is performed close ADAL Item 101.</p>		
52	EO-AOA-55-14	<p>AERODYNAMIC SEAL LH UPPER HORIZONTAL STABILIZER NO. 1 REPLACEMENT</p> <p>Perform replacement of the Aerodynamic Seal LH Upper Horizontal Stabilizer No. 1 which is partially missing with new one iaw AMM 55-14-27-000-801-A and AMM 55-14-27-400-801-A.</p> <p>NOTE: After completion update ADAL item 118.</p>		
53	EO-AOA-55-15	<p>INSTALLATION OF DIMPLED WASHER IN VERTICAL STABILIZER AERODYNAMIC SEALS ATTACHING PARTS - MODIFICATION</p> <p>Perform installation of dimpled washer in vertical stabilizer aerodynamic seals attaching parts iaw SB190-55-0006, 3.Accomplishment Instructions.</p> <p>NOTE: Aircraft is applicable to GROUP 2.</p> <p>NOTE: Perform rework the existing vertical stabilizer aerodynamic seals.</p> <p>NOTE for planning: Sheet metal mechanic is required.</p> <p>NOTE for engineering: After accomplishment of this EO, make accomplishment note in:</p> <ul style="list-style-type: none"> - SB status; - Modification log book; - and inform Embraer that SB has been incorporated. 		



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54	EO-AOA-57-11	<p>WINGS - CLIPS OF FLAP SHROUD PANEL AND MULTIFUNCTION SPOILER FITTINGS REPLACEMENT - LH SIDE</p> <p>Perform SB 190-57-0049 iaw Item 3. ACCOMPLISHMENT INSTRUCTIONS Parts: A, B, C, D and E.</p> <p>NOTE for engineering: After accomplishment of this modification make note in aircraft log book: `SB190-57-0049 performed by , @.....fh, fc, (date).`</p>		
55	EO-AOA-57-12	<p>WINGS - CLIPS OF FLAP SHROUD PANEL AND MULTIFUNCTION SPOILER FITTINGS REPLACEMENT - RH SIDE</p> <p>Perform SB 190-57-0049 iaw Item 3. ACCOMPLISHMENT INSTRUCTIONS Parts: A, B, C, D and F.</p> <p>NOTE for engineering: After accomplishment of this modification make note in aircraft log book: `SB190-57-0049 performed by , @.....fh, fc, (date).`</p>		
56	EO-AOA-57-13	<p>WINGS-STRUCTURAL REINFORCEMENT OF THE WING STUB LOWER SKIN AT AC AUXILIARY BOOST-PUMP CANISTER SATELLITE HOLES-MODIFICATION.</p> <p>Perform modification of structural reinforcement of the wing stub lower skin at AC Auxiliary boost-pump canister satellite holes iaw SB 190-57-0039 R01-3. Accomplishment Instruction and SB 190-57-0039 IN1.</p> <p>NOTE for Planning: Sheet metal specialist is needed.</p> <p>NOTE for Engineering: After accomplishment of this EO, make a record in Modification log book and SB status list.</p>		
57	Z250-002	<p>GVI - CENTER FUSELAGE III, AREA COVERED BY GALLEY AND TOILET, AND AREA BEHIND GALLEY AND TOILET TO REAR PRESSURE BULKHEAD FWD FACE - INTERNAL</p> <p>General Visual Inspection of Center Fuselage III, area covered by Galley and Toilet, area behind Galley and Toilet to Rear Pressure Bulkhead FWD Face and area below Floor Panels - Internal Side of Fuselage with Floor Panels Removal.</p> <p>NOTE: EWIS. Galley and Toilet have to be removed.</p> <p>NOTE: Interval is 15000 fh.</p>		